

Kepler Problem – PHY 4241 Notes

Bernd A. Berg^a

^{a)} *Department of Physics, Florida State University, Tallahassee, FL 32306-4350*

(Dated: September 3, 2017)

Kepler problem as discussed in class.

PACS numbers:

I. SOLUTION OF THE KEPLER PROBLEM FOR NUMERICAL IMPLEMENTATION

The purpose of this section is to cast the analytical solution of the Kepler problem into a form, which allows for easy implementation into a computer program (here done in Fortran). The final solutions are translated back to the initially given inertial system. There are many treatments of the Kepler problem, for instance [1, 2]. We follow to some extent by Landau and Lifschiz.

A. Central potential problem

Let t be the time and the masses be m_1 and m_2 . In a frame Σ defined by Cartesian orthonormal vectors $\hat{e}_1, \hat{e}_2, \hat{e}_3$ positions and velocities of the masses m_i , ($i = 1, 2$) are given by:

$$\vec{r}_i(t) = \begin{pmatrix} x_i^1(t) \\ x_i^2(t) \\ x_i^3(t) \end{pmatrix} \quad \text{and} \quad \vec{v}_i(t) = \begin{pmatrix} \dot{x}_i^1(t) \\ \dot{x}_i^2(t) \\ \dot{x}_i^3(t) \end{pmatrix} \quad (1)$$

where the dot denotes the time derivative. For positions and velocities the initial conditions at time t_0 are

$$\vec{r}_i(t_0) \quad \text{and} \quad \vec{v}_i(t_0), \quad i = 1, 2 \quad (2)$$

in the frame Σ . The computer program will calculate $\vec{r}_i(t)$ and $\vec{v}_i(t)$, $i = 1, 2$ for any desired time t .

We denote the total mass by

$$M = m_1 + m_2, \quad (3)$$

the Center of Mass (CM) position and velocity are defined by

$$\vec{r}_{\text{cm}}(t) = \frac{m_1 \vec{r}_1(t) + m_2 \vec{r}_2(t)}{M}, \quad (4)$$

$$\vec{v}_{\text{cm}}(t) = \frac{m_1 \vec{v}_1(t) + m_2 \vec{v}_2(t)}{M}. \quad (5)$$

Due to momentum conservation one finds

$$\vec{v}_{\text{cm}}(t) = \vec{v}_{\text{cm}}(t_0) \quad \text{and} \quad (6)$$

$$\vec{r}_{\text{cm}}(t) = \vec{r}_{\text{cm}}(t_0) + \vec{v}_{\text{cm}}(t_0) \Delta t, \quad \Delta t = (t - t_0). \quad (7)$$

Defining the difference coordinates and velocities by

$$\vec{r}_{12}(t) = \vec{r}_1(t) - \vec{r}_2(t), \quad (8)$$

$$\vec{v}_{12}(t) = \vec{v}_1(t) - \vec{v}_2(t), \quad (9)$$

the positions of the particles are

$$\vec{r}_1(t) = \vec{r}_{\text{cm}}(t) + \frac{m_2 \vec{r}_{12}(t)}{M}, \quad (10)$$

$$\vec{r}_2(t) = \vec{r}_{\text{cm}}(t) - \frac{m_1 \vec{r}_{12}(t)}{M}. \quad (11)$$

These equations show that the CM is located in between the particles on the straight line connecting them. Equations for the velocities are defined by taking the time derivatives.

So, the task has become to solve the equations of motion for the difference coordinates and to implement the solution numerically. In the following we used the notation

$$r_{12} = |\vec{r}_{12}(t)|. \quad (12)$$

For a central potential $U(r_{12})$ the energy

$$E = \frac{m_1 \vec{v}_1(t)^2}{2} + \frac{m_2 \vec{v}_2(t)^2}{2} + U(r_{12}) \quad (13)$$

is conserved, i.e., does not depend on the time t .

As the difference coordinate $\vec{r}_{12}(t)$ stays invariant under transformation to the CM frame Σ' defined by

$$\vec{r}'_{\text{cm}}(t) = 0. \quad (14)$$

We can calculate its time dependence in the CM system and still use Eq. (10) and (11) to find the time dependence of the original coordinates. In the CM systems these equations simplify to

$$\vec{r}'_1(t) = \frac{m_2 \vec{r}'_{12}}{M} \quad \text{and} \quad \vec{r}'_2(t) = -\frac{m_1 \vec{r}'_{12}}{M}. \quad (15)$$

In the following we drop the primes and continue to work in the CM frame (we shall use the primes for yet another frame soon).

Inserting the derivatives of equation (15) (without primes) into the energy conservation (13), we find

$$E_{\text{cm}} = \frac{\mu \vec{v}_{12}(t)^2}{2} + U(r_{12}) \quad \text{where} \quad \mu = \frac{m_1 m_2}{M}. \quad (16)$$

μ is called reduced mass. To simplify the notation, we drop the subscript 12:

$$\vec{r}(t) = \vec{r}_{12}(t) \quad \text{and} \quad \vec{v}(t) = \vec{v}_{12}(t). \quad (17)$$

With $\vec{r}(t_0)$ and $\vec{v}(t_0)$ given, we want to find $\vec{r}(t)$ for the central potential problem (16). Besides momentum and energy the angular momentum is conserved:

$$\vec{L} = \vec{r}_1 \times \vec{p}_1 + \vec{r}_2 \times \vec{p}_2 \quad (18)$$

where $\vec{p}_i = m_i \vec{v}$, $i = 1, 2$ are the momenta of the masses. Using (15) we have

$$\vec{L} = \mu \vec{r}(t) \times \vec{v}(t) = \mu \vec{r}(t_0) \times \vec{v}(t_0). \quad (19)$$

We deal with $\vec{L} = 0$ later. Assuming $\vec{L} \neq 0$, the motion takes place in the plane spanned by $\vec{r}(t_0)$ and $\vec{v}(t_0)$ and it is convenient to describe it in a new coordinate frame Σ' defined by Cartesian orthonormal vectors $\hat{e}'_1, \hat{e}'_2, \hat{e}'_3$ with

$$\hat{e}'_1 = \frac{\vec{r}(t_0)}{|\vec{r}(t_0)|}, \quad (20)$$

$$\hat{e}'_2 = \frac{\vec{v}(t_0) - [\vec{v}(t_0) \cdot \hat{e}'_1] \hat{e}'_1}{|\vec{v}(t_0) - [\vec{v}(t_0) \cdot \hat{e}'_1] \hat{e}'_1|}, \quad (21)$$

$$\hat{e}'_3 = \hat{L} = \frac{\vec{L}}{L} \text{ with } L = |\vec{L}|. \quad (22)$$

By definition of the angular momentum this is a right-handed frame. In this frame the absolute value of the angular momentum reads

$$L = \mu r(t)^2 \dot{\phi}'(t), \quad (23)$$

where the azimuth angle ϕ' is defined by

$$x'(t) = r(t) \cos[\phi'(t)], \quad (24)$$

$$y'(t) = r(t) \sin[\phi'(t)]. \quad (25)$$

Defining the effective potential by

$$U_{\text{eff}}(r) = U(r) + \frac{L^2}{2\mu r(t)^2}, \quad (26)$$

where $L^2/[2\mu r(t)^2]$ is called centrifugal energy, the energy conservation (16) becomes

$$E_{\text{cm}} = \frac{\mu}{2} \dot{r}(t)^2 + U_{\text{eff}}(r). \quad (27)$$

Separation of variables yields

$$dt = dr \sqrt{\frac{\mu}{2[E_{\text{cm}} - U_{\text{eff}}(r)]}},$$

$$\Delta t = t - t_0 = \int_{r(t_0)}^{r(t)} dr \sqrt{\frac{\mu}{2[E_{\text{cm}} - U_{\text{eff}}(r)]}}. \quad (28)$$

To get $\phi'(t)$ we use (23):

$$d\phi' = \frac{L dt}{\mu r^2} = \frac{L dr}{r^2 \sqrt{2\mu[E_{\text{cm}} - U_{\text{eff}}(r)]}}$$

$$\Delta\phi' = \phi'(t) - \phi'(t_0) \quad (29)$$

$$= \int_{r(t_0)}^{r(t)} \frac{L dr}{r^2 \sqrt{2\mu[E_{\text{cm}} - U_{\text{eff}}(r)]}},$$

where due to the choice of the Σ' coordinate frame of Eq. (20) to (22) we have

$$\phi'(t_0) = 0. \quad (30)$$

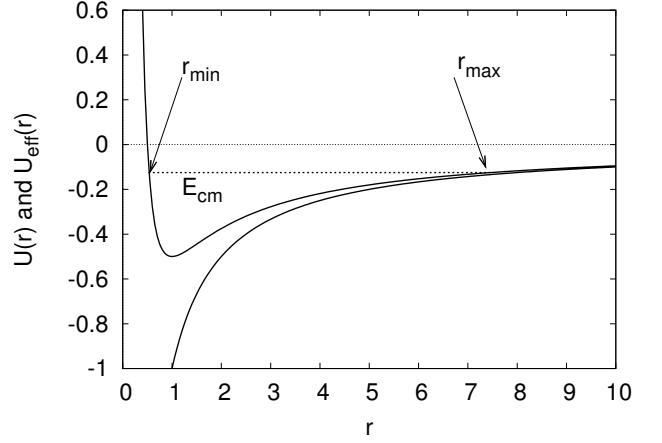


FIG. 1: Potential and effective potential for the Kepler problem.

B. Kepler problem

We specialize now to the gravitational potential

$$U(r) = -\frac{\alpha}{r} \text{ with } \alpha = G m_1 m_2. \quad (31)$$

The effective potential becomes

$$U_{\text{eff}}(r) = -\frac{\alpha}{r} + \frac{L^2}{2\mu r(t)^2}. \quad (32)$$

An example is shown in Fig. 1. For $E_{\text{cm}} \geq 0$ the distance between the masses escapes to infinity, while for $E_{\text{cm}} < 0$ it is confined between r_{min} and r_{max} . There are no solutions for $E_{\text{cm}} < U_{\text{eff}}^{\text{min}}$ with

$$U_{\text{eff}}^{\text{min}} = U_{\text{eff}}(r_{\text{min}}^0) = -\frac{\alpha^2}{2} \frac{\mu}{L^2}, \quad r_{\text{min}}^0 = \frac{L^2}{\alpha\mu}. \quad (33)$$

At r_{min}^0 the orbit is a circle. For (31) the solution of the integral (29) is elementary

$$\phi'(t) = \arccos\left(\frac{L/r(t) - \mu\alpha/L}{\sqrt{2\mu E_{\text{cm}} + \mu^2\alpha^2/L^2}}\right) - \arccos\left(\frac{L/r(t_0) - \mu\alpha/L}{\sqrt{2\mu E_{\text{cm}} + \mu^2\alpha^2/L^2}}\right) \quad (34)$$

We map on the coordinate conventions of the literature [1, 2] in which the r_{min} position, called *perihelion* or *pericenter*, is at $\phi = 0$. This defines the the azimuth angle

$$\phi(t) = \arccos\left(\frac{L/r(t) - \mu\alpha/L}{\sqrt{2\mu E_{\text{cm}} + \mu^2\alpha^2/L^2}}\right), \quad (35)$$

so that

$$\phi'(t) = \phi(t) - \phi(t_0) \quad (36)$$

holds for the previously introduced $\phi'(t)$. The angle $\phi(t)$ is defined with respect to a coordinate system, which we call Σ_1 (1 for literature). It is rotated with respect to Σ' by $\phi(t_0)$ about the common \hat{e}'_3 axis.

We can write Eq. (35) as

$$\cos[\phi(t)] = \frac{L/r(t) - \mu\alpha/L}{(\mu\alpha/L)\sqrt{1 + 2E_{\text{cm}}L^2/(\mu\alpha^2)}}. \quad (37)$$

With the definitions of the parameter p , $2p$ is called the *latus rectum*, and the *eccentricity* e ,

$$p = \frac{L^2}{\mu\alpha} \quad \text{and} \quad e = \sqrt{1 + \frac{2E_{\text{cm}}L^2}{\mu\alpha^2}}, \quad (38)$$

the orbit in Σ_1 becomes

$$\frac{p}{r(t)} = 1 + e \cos[\phi(t)] \quad \text{or} \quad r(t) = \frac{p}{1 + e \cos[\phi(t)]}. \quad (39)$$

This is a conic section with the coordinate origin in a focal point. The shortest distance from the focal point is

$$r_{\min} = \frac{p}{1 + e}, \quad (40)$$

which corresponds to the pericenter. For $U_{\text{eff}}^{\min} < E_{\text{cm}} < 0$ the orbit is an ellipse and the largest distance from the focal point is

$$r_{\max} = \frac{p}{1 - e}, \quad (41)$$

which corresponds to the *apocenter*. The *turning points* r_{\min} and r_{\max} of the orbit are also called *apsides*. They are indicated in Fig. 1. The large half-axis is

$$a = \frac{p}{1 - e^2} = \frac{\alpha}{2|E_{\text{cm}}|} \quad (42)$$

and the small half-axis

$$b = \frac{p}{\sqrt{1 - e^2}} = \frac{L}{\sqrt{2\mu|E_{\text{cm}}|}}. \quad (43)$$

The ellipse becomes a circle for $e = 0$ ($E_{\text{cm}} = U_{\text{eff}}^{\min}$). The position and velocity vectors will be orthogonal $\vec{r} \cdot \vec{v} = 0$, and we have a relation between their magnitudes. With

$$r_0 = r(t_0) = |\vec{r}(t_0)|, \quad (44)$$

$$v_0 = v(t_0) = |\vec{v}(t_0)|, \quad (45)$$

the relation $E_{\text{cm}} = U_{\text{eff}}^{\min}$ yields:

$$\frac{\mu}{2}(v_0)^2 - \frac{\alpha}{r_0} = -\frac{\alpha^2}{2\mu(r_0)^2(v_0)^2}, \quad (46)$$

$$(v_0)^4 - \frac{2\alpha}{\mu r_0}(v_0)^2 + \frac{\alpha^2}{\mu^2(r_0)^2} = 0. \quad (47)$$

The argument of the $\pm\sqrt{}$ part of the solution turns out to be zero, so that we end up with the unique result

$$v_0 = \sqrt{\frac{\alpha}{\mu r_0}} \quad (48)$$

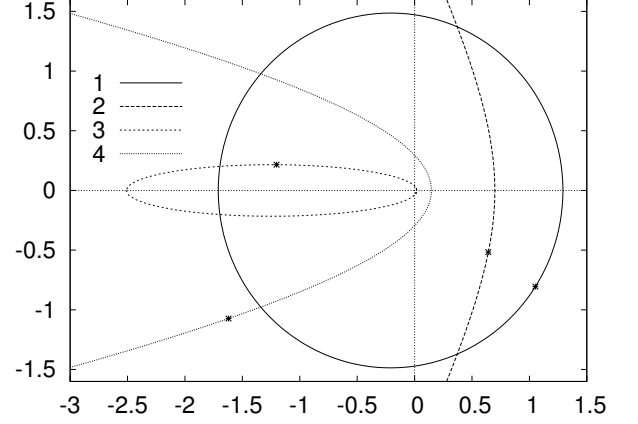


FIG. 2: Elliptical and hyperbolic orbits corresponding to the initial conditions of table I.

and $\vec{v}_0 \cdot \vec{r}_0 = 0$.

For $E_{\text{cm}} > 0$ ($e > 1$) the orbits are hyperbolic and escape to infinity for the two solutions for the equation $1 + e \cos(\phi_\infty) = 0$. For $E_{\text{cm}} = 0$ the orbit is parabolic and $\phi_\infty = \pi$.

#	i	m_i	Initial Positions			Initial Velocities		
			$x_{i,0}^1$	$x_{i,0}^2$	$x_{i,0}^3$	$\dot{x}_{i,0}^1$	$\dot{x}_{i,0}^2$	$\dot{x}_{i,0}^3$
1	1	0.651	0.585	-0.238	-0.755	-0.828	-0.865	-0.726
	2	0.931	-0.096	0.000	0.357	-0.209	0.107	-0.660
2	1	1.510	0.460	-0.359	-0.234	-0.918	-0.941	-0.323
	2	0.126	-0.066	-0.090	-0.809	0.789	0.788	0.620
3	1	1.328	-0.125	0.898	0.194	-0.452	0.172	0.125
	2	1.999	-0.449	-0.085	-0.454	-0.976	-0.990	-0.968
4	1	0.180	0.204	-0.968	-0.753	-0.811	-0.632	0.784
	2	1.560	-0.889	-0.979	0.854	-0.323	-0.774	-0.533

TABLE I: Examples of initial conditions corresponding to orbits shown in Fig. 2.

In Fig. 2 we give examples of orbits in the Σ_1 frame, which correspond to the initial conditions and velocities compiled in table I. Crosses in the figure indicate the corresponding initial positions in the Σ_1 frame.

-
- [1] L.D. Landau and E.M. Lifschiz, *Lehrbuch der Theoretischen Physik I*, Akademie Verlag, Berlin 1962.
 - [2] J.B. Marion and S.T. Thornton *Classical Dynamics of Particles and Systems*, Harcourt Brace & Company, Orlando, FL 1995.